

Orbit stability at SPEAR2 and plans for SPEAR3



J. Safranek for the SSRL Accelerator Group

○ SPEAR2

- ↪ BPM and orbit feedback performance
- ↪ Present orbit stability
- ↪ Improvements in LCW and air temperature

○ SPEAR3

- ↪ Stability specifications
- ↪ Mitigating sources of orbit motion
- ↪ Orbit measurement/feedback
- ↪ Application programs for orbit control

SPEAR/SSRL



- **1972 – SPEAR built (colliding beams)**
- **1974 – First beamline**
- **1982 – First fast orbit feedback**
- **1990 – SPEAR dedicated solely to S.R., with independent injector**
- **2002**
 - ↪ **11 beamlines, operations ~9 months/year**
 - ↪ **$e_x = 140$ nm, $I_{max} = 100$ mA, $E = 3$ GeV**
- **2003**
 - ↪ **SPEAR3 – complete accelerator rebuild**
 - ↪ **$e_x = 18$ nm, $I_{max} = 500$ mA, $E = 3$ GeV**

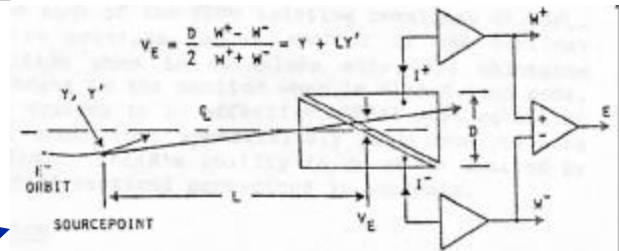
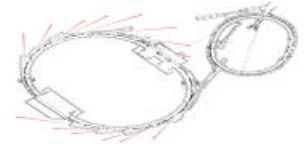


Fig. 1. Split anode vertical position monitor system. The synchrotron beam trajectory is tangential to the electron orbit at the sourcepoint.

SPEAR2 BPM performance



- **J.L. Pellegrin electronics, 1970's.**
- **Peak detector designed for single bunch**
 - ↪ **Requires “mondo” bucket**
 - ↪ **Significant fill pattern dependence**
 - ↪ **Low intensity dependence; AGC loop - 1 dB attenuator steps**
- **10-20 mm rms noise with 3.8 seconds/orbit**
 - ↪ **Multiplexed to single ADC**
 - ↪ **1250 reads/orbit**
- **30 BPMs ($n_x = 7.15$, $n_y = 5.24$)**

SPEAR2 orbit feedback



○ Slow orbit feedback (J. Corbett et al.)

↪ 50 second cycle time

↪ SVD orbit correction using e- BPMs only

○ Fast feedback on photon monitors (R. Hettel)

↪ ~60 Hz bandwidth

↪ Uses local electron orbit bumps and single photon monitors

↪ Steering magnets – coils on quadrupoles, Al chamber (50 Hz pole)

○ Mirror feedback (T. Rabedeau)

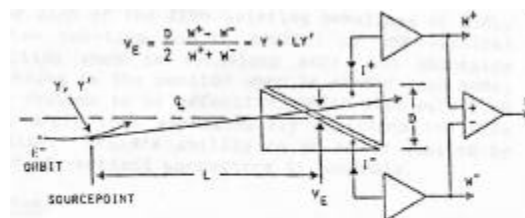
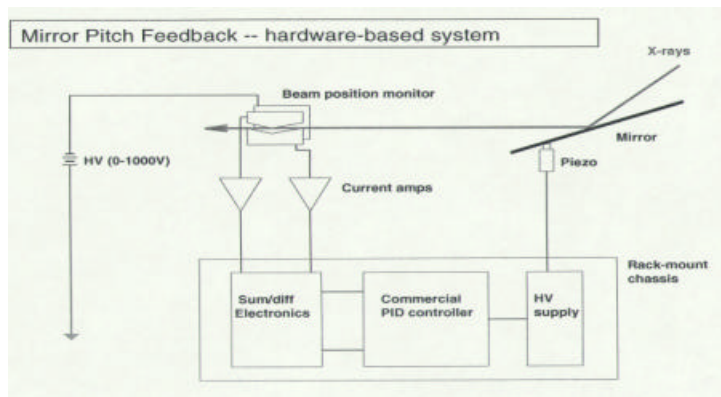
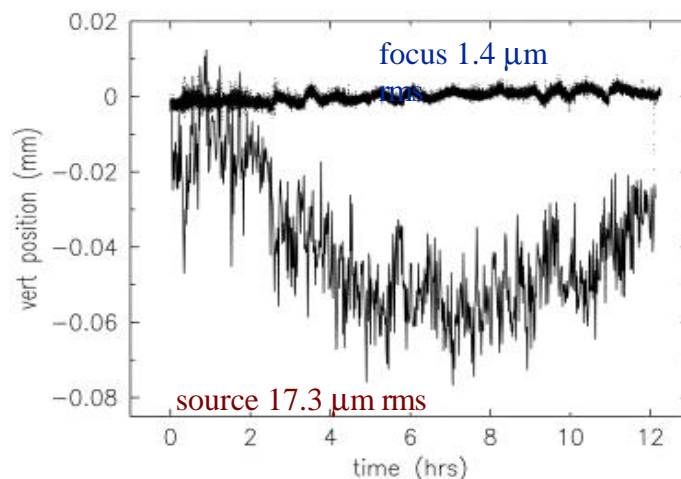
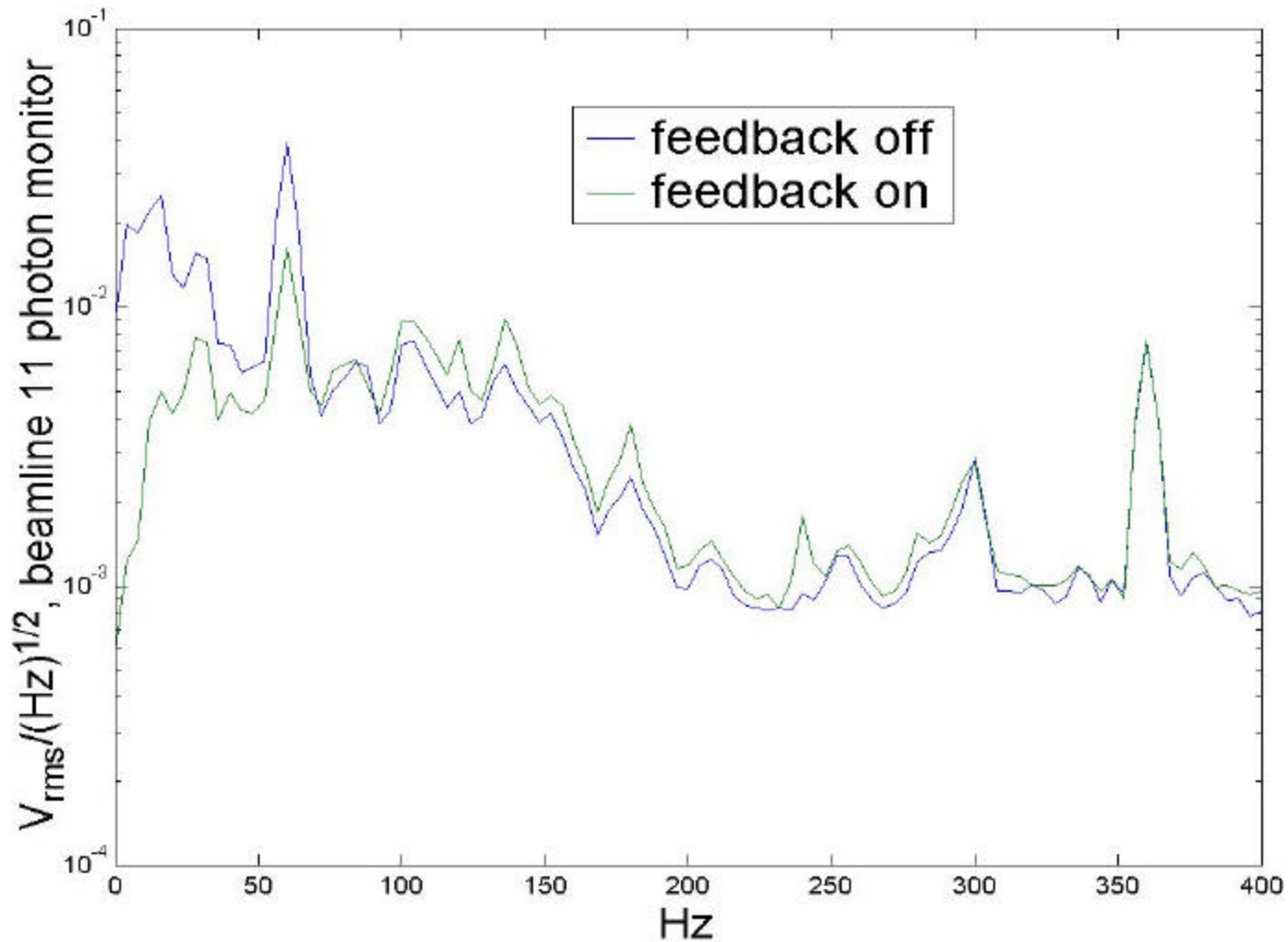
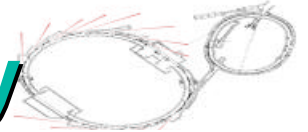


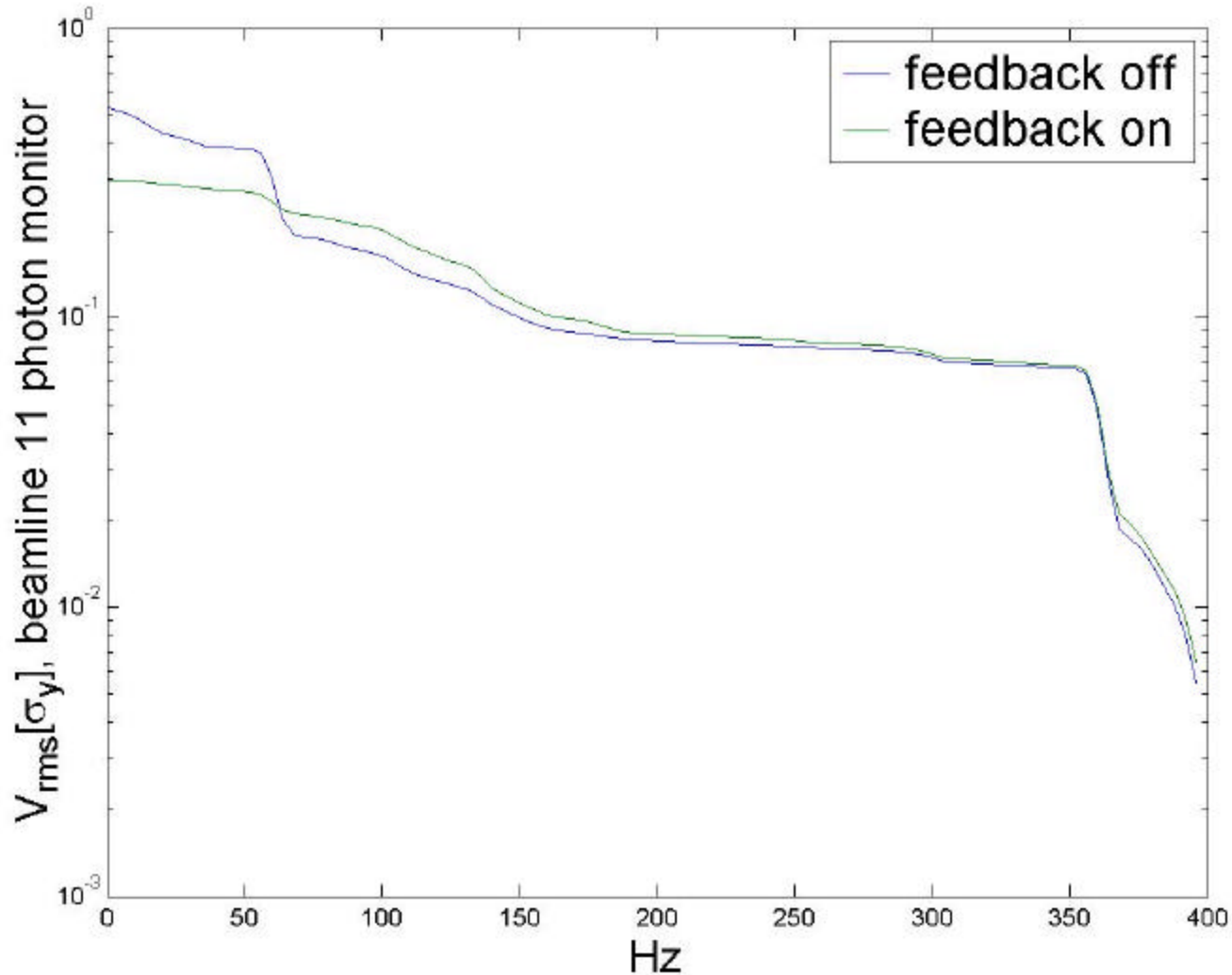
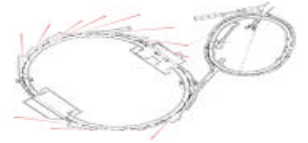
Fig. 1. Split anode vertical position monitor system. The synchrotron beam trajectory is tangential to the electron orbit at the sourcepoint.



SPEAR2 power spectral density



SPEAR2 PSD Integral



Orbit stability over two weeks



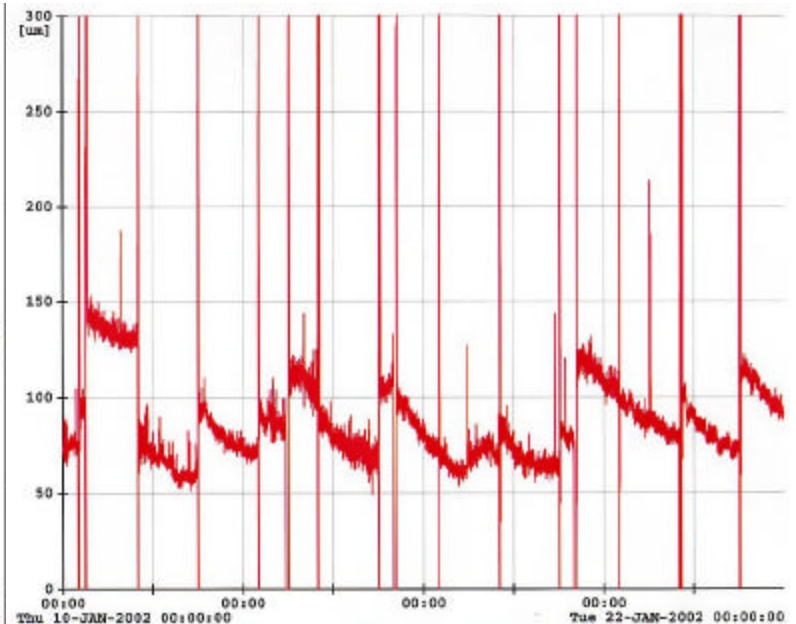
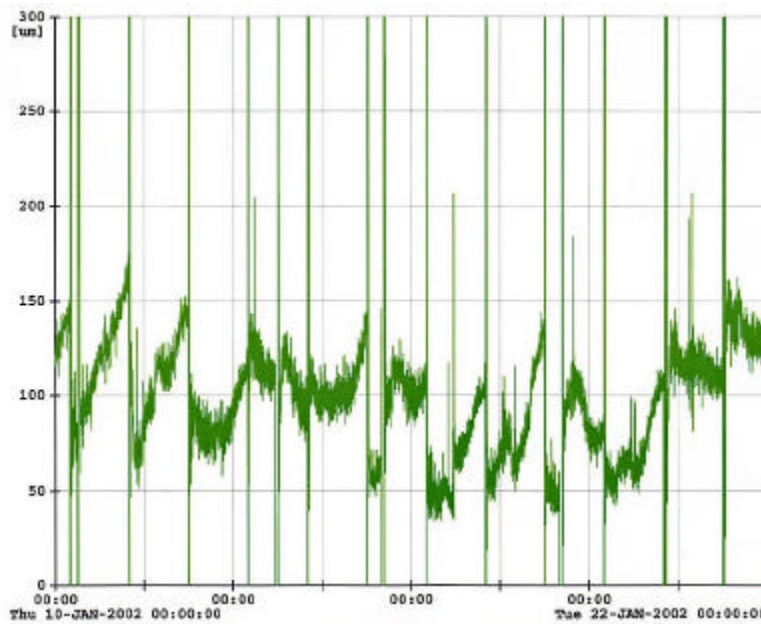
X_rms

Y_rms

300 mm

150 mm

0 mm



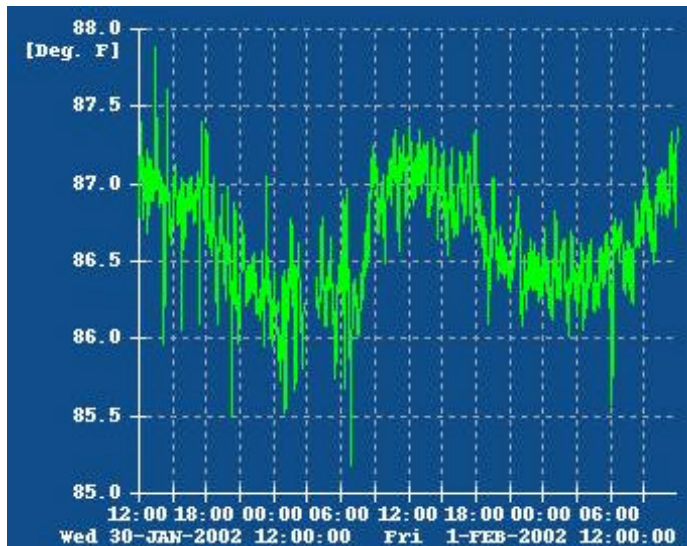
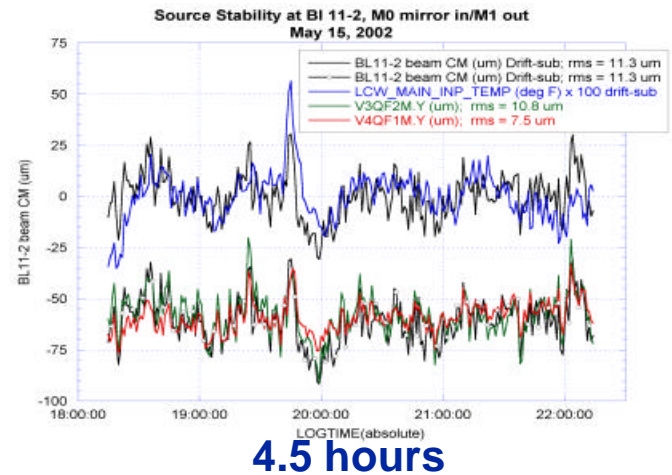
Cooling water temperature



LCW variations effect beamlines

rms_dT,C	weekly	daily	6 hours
1997	.094	.069	.052
1998	.118	.078	.052
1999	.138	.068	.057
2000	.107	.102	.066
2001	.185	.163	.103

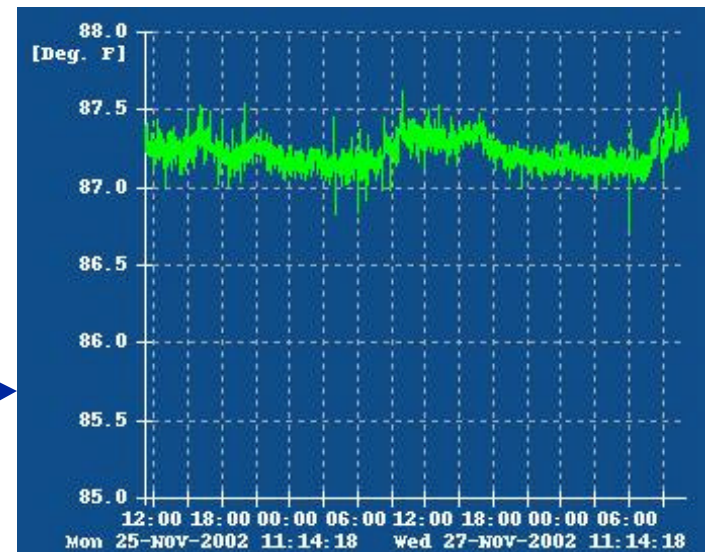
11-2 CM, LCW



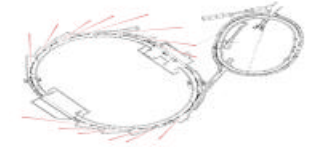
LCW temperature variations reduced

← Jan'02

Nov'02 →

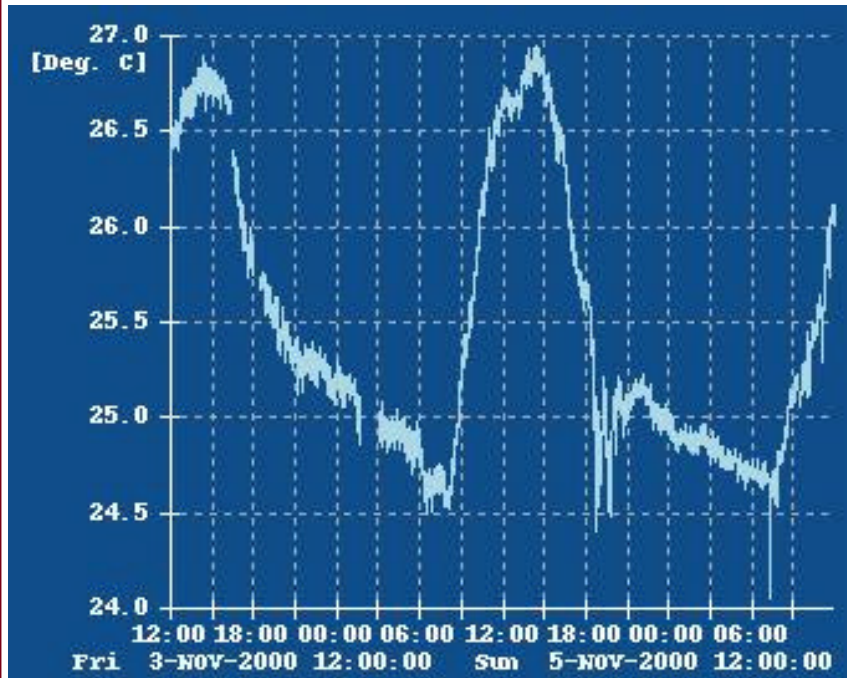


Air temperature stability

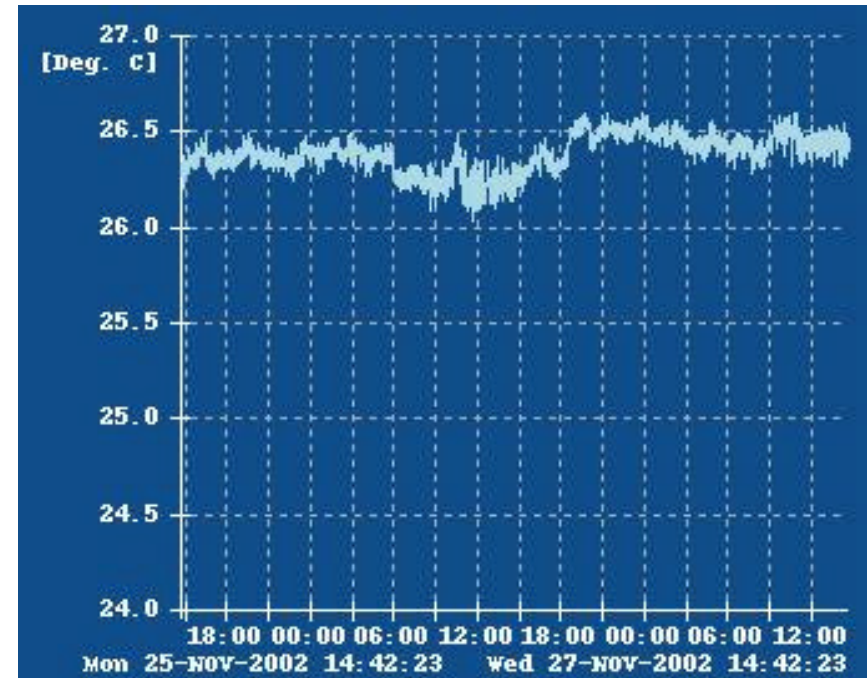


Extending concrete tunnel reduced air temperature variations.

November, 2000



November, 2002



SPEAR3 orbit stability specifications



Table 3.23 RMS source point beam dimensions and stability requirements for SPEAR 3 (rms, 1% coupling).

		Electron	Photon	10%
Dipole	σ_x (μm)	160	160	16
	$\sigma_{x'}$ (μrad)	236	mrads	< mrad
	σ_y (μm)	51	51	5
	$\sigma_{y'}$ (μrad)	11	136	14
ID-Wiggler	σ_x (μm)	435	435	43
	$\sigma_{x'}$ (μrad)	43	2-20 mrad	< mrad
	σ_y (μm)	30	30	3 [†]
	$\sigma_{y'}$ (μrad)	6	136	14
ID-100 per und.	σ_x (μm)	435	435	43
	$\sigma_{x'}$ (μrad)	43	43	4
	σ_y (μm)	30	30	3 [†]
	$\sigma_{y'}$ (μrad)	6	15	1.5

†. This requirement can be relaxed to 5 μm due to 50 μm minimum vertical spot size achieved from present focusing mirrors.

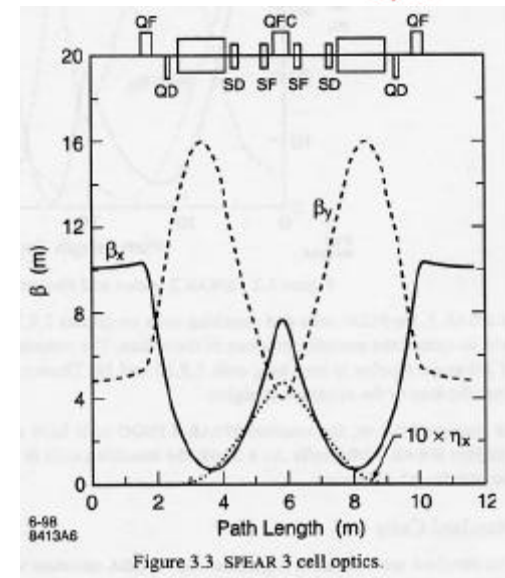


Figure 3.3 SPEAR 3 cell optics.

E	3 GeV
I	500 mA
e_x	18 nm
n_x	14.19
n_y	5.23
n_s	.008
S_E	.097%
V_{rf}	3.2 MV
a_c	.0011

SPEAR3 alignment

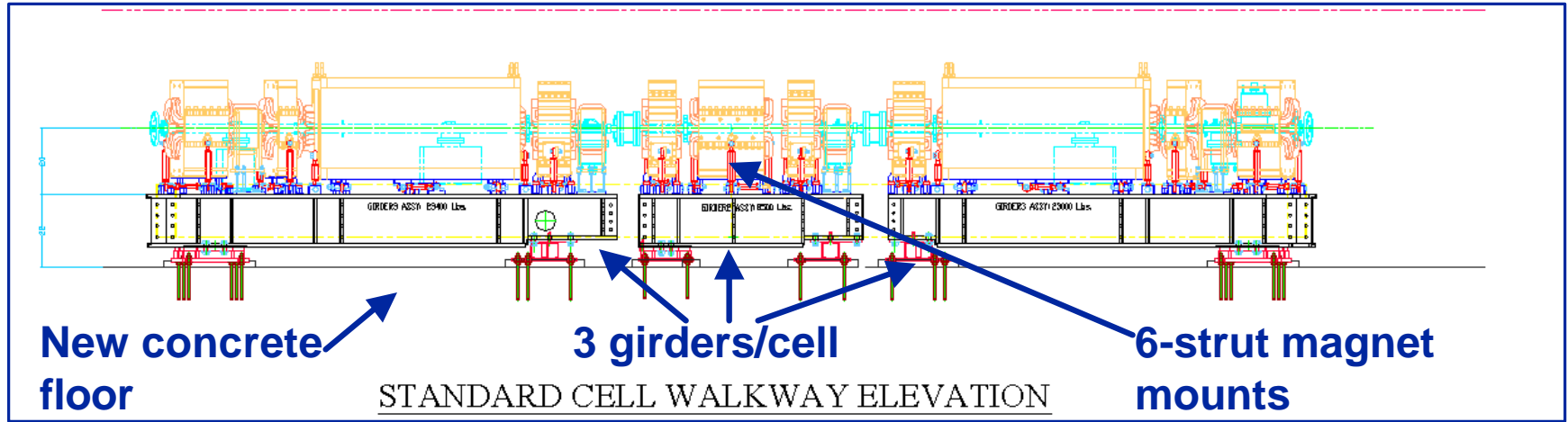
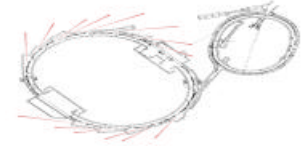


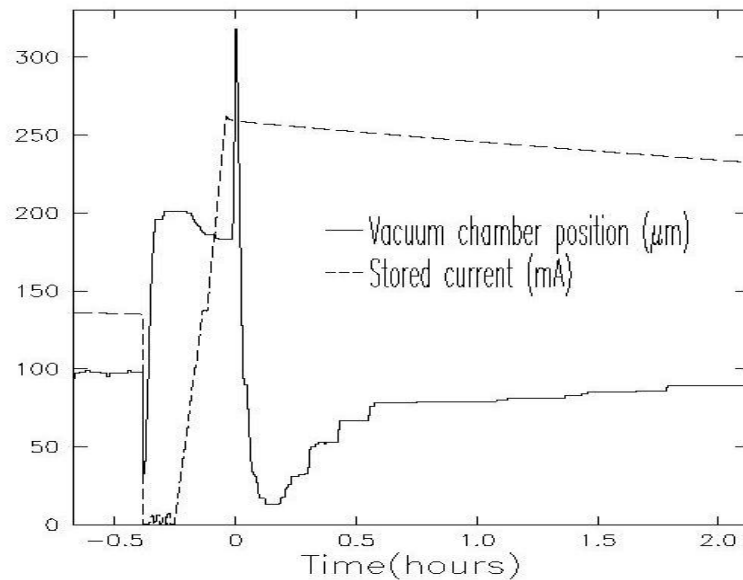
Table 4.28 Tolerances for lattice-component alignment

Component	Rotation rms error (mrad)			Displacement rms error (mm)		
	Θ_Z	Θ_X	Θ_Y	ΔX	ΔY	ΔZ
	roll	pitch	yaw	D_x	D_y	D_s
Bend magnets	0.50	0.30	0.50	0.20	0.20	1.50
Quadrupoles	0.50	0.50	0.50	0.20	0.20	1.50
Sextupoles	0.50	0.20	1.00	0.20	0.20	1.50
RF cavity	1.00	0.50	0.50	0.20	0.20	1.50
bpm	1.00	0.50	0.50	0.25	0.25	1.50
Vacuum Chamber	--	--	--	0.25	0.25	0.50
Kicker	1.00	0.25	0.25	0.50	0.50	1.50
Insertion Device	0.50	--	--	0.25	0.25	1.50

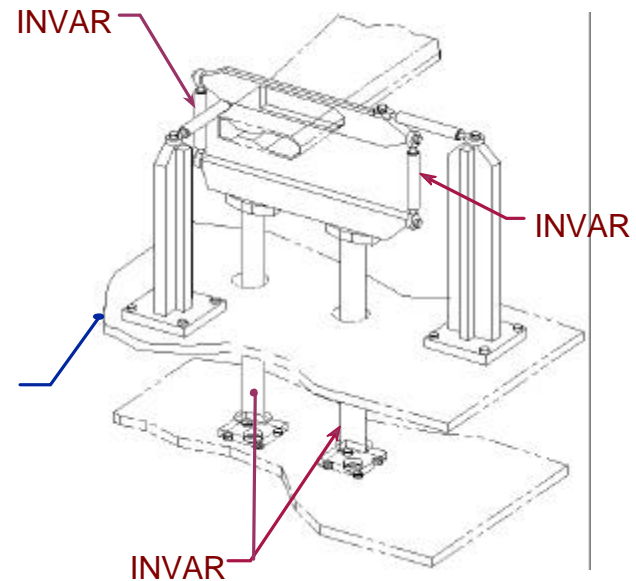
SPEAR3 vacuum chamber, BPM stability



- All synchrotron radiation hits H₂O-cooled masks.
- BPM mechanical motion constrained by Invar struts.



NSLS chamber motion



SPEAR 3 chamber/BPM supports

3 μm/°C vert, 15 μm/°C hor

Magnet vibrations



- **Allowable uncorrelated magnet motion (10% beam size)**

- ↪ **Specification: 33 nm vertical, 250 nm horizontal**

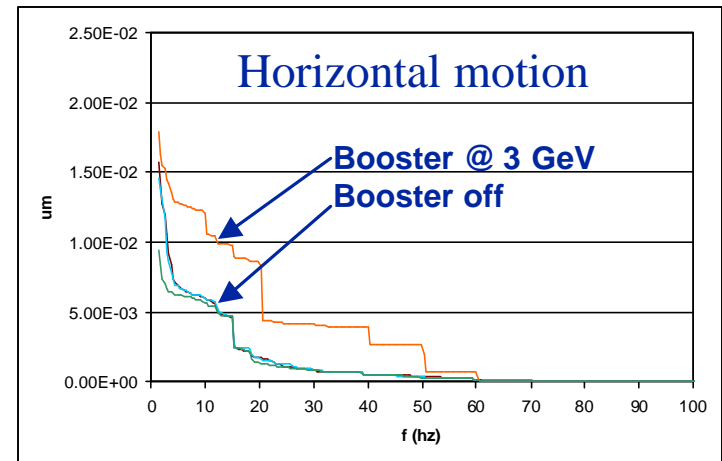
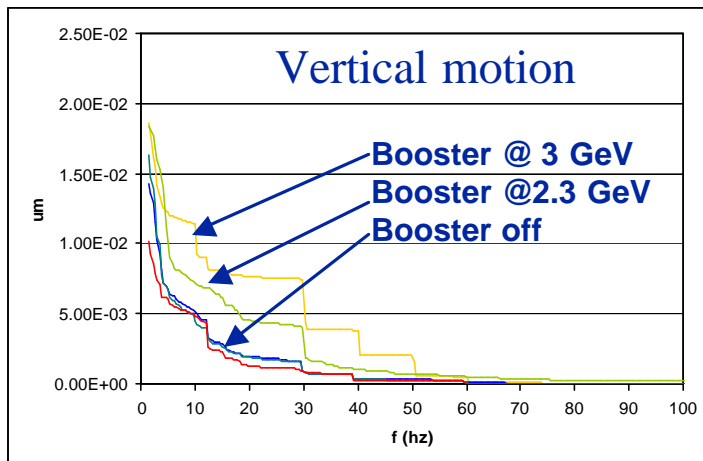
- ↪ **Measured in SPEAR2: 70 nm vertical, 700 nm horizontal**

- **Lowest vibration mode**

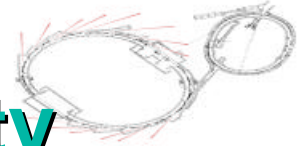
- ↪ **SPEAR2: 4.5 Hz**

- ↪ **SPEAR3: 17 Hz**

- **Ground motion measurements (A. Seryi)**



SPEAR3 power supply stability



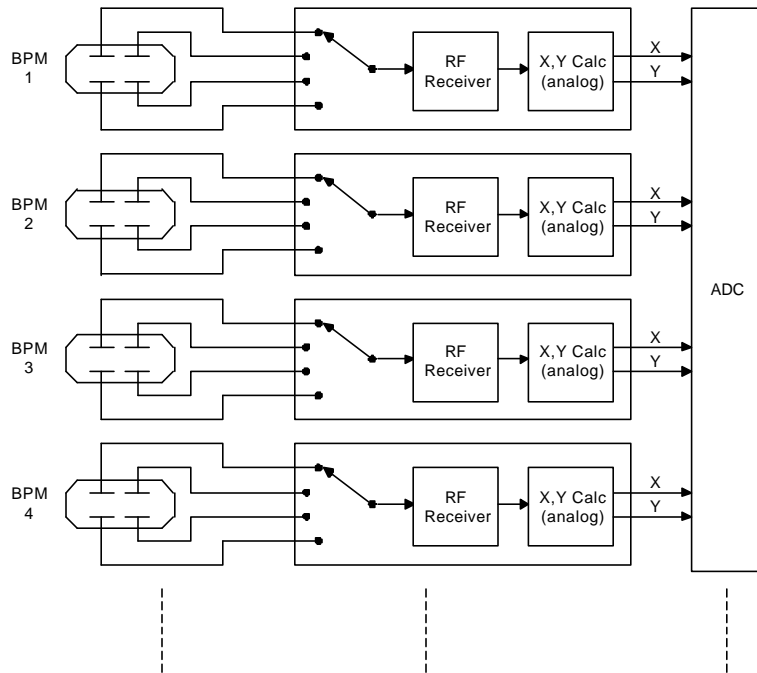
- **Pulse width modulation (20-40 kHz)**
- **Dipole and quad supply bandwidth > 100 Hz (suppression of line voltage transients)**
- **24-hour stability specs (supplies exceed specs):**
 - ↪ **Dipole: 50 PPM**
 - ↪ **Quadrupoles: 100 PPM**
 - ↪ **Sextupoles: 500 PPM**
 - ↪ **Correctors: 500 PPM**

SPEAR3 BPM Processors



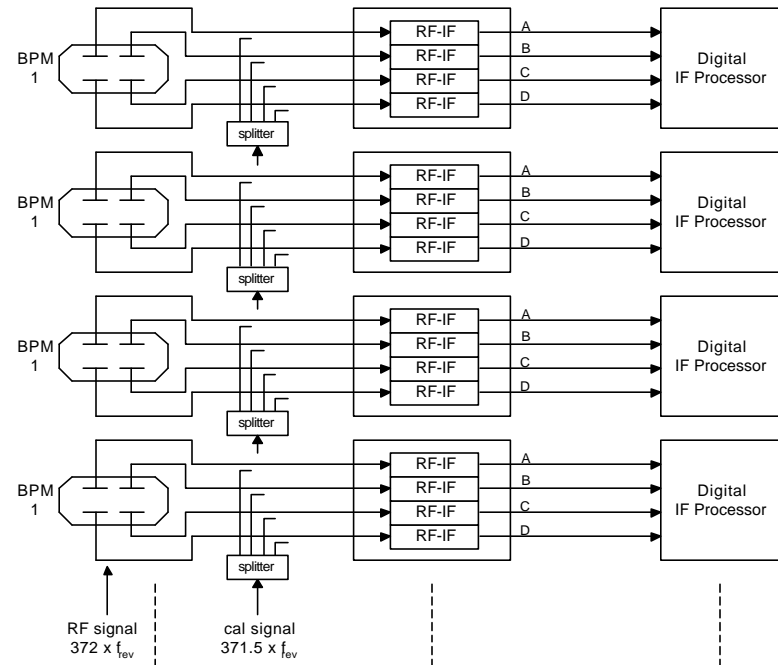
4:1 button MUX (Bergoz) analog receiver

multi-turn BPM measurement
~100-200 Hz BW



parallel processing digital receiver (Echotek)

1st turn/single-turn/multi-turn BPM
measurement
~1 μm at 4 kHz

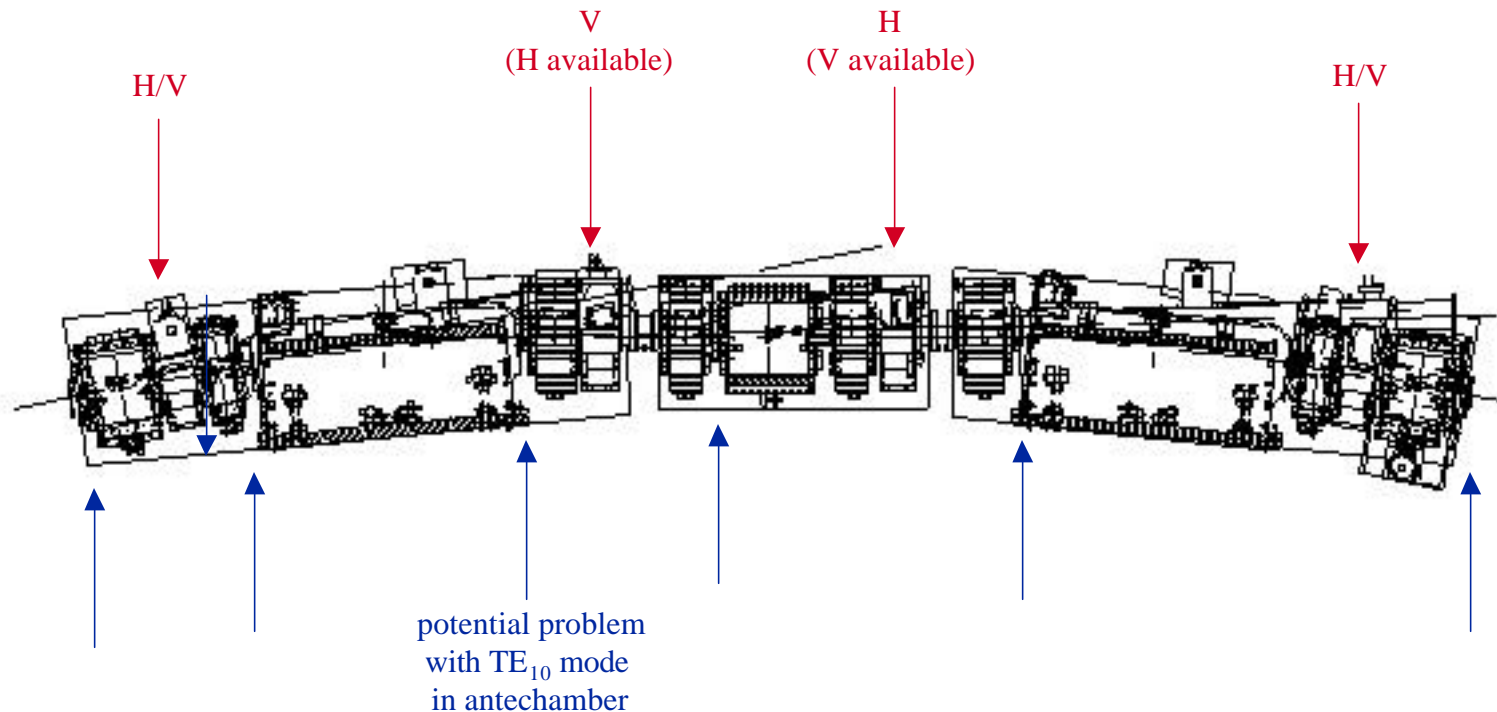


SPEAR3 BPM/Corrector locations



Corrector locations

(72 total; 54 H and 54 V correctors used for orbit feedback)



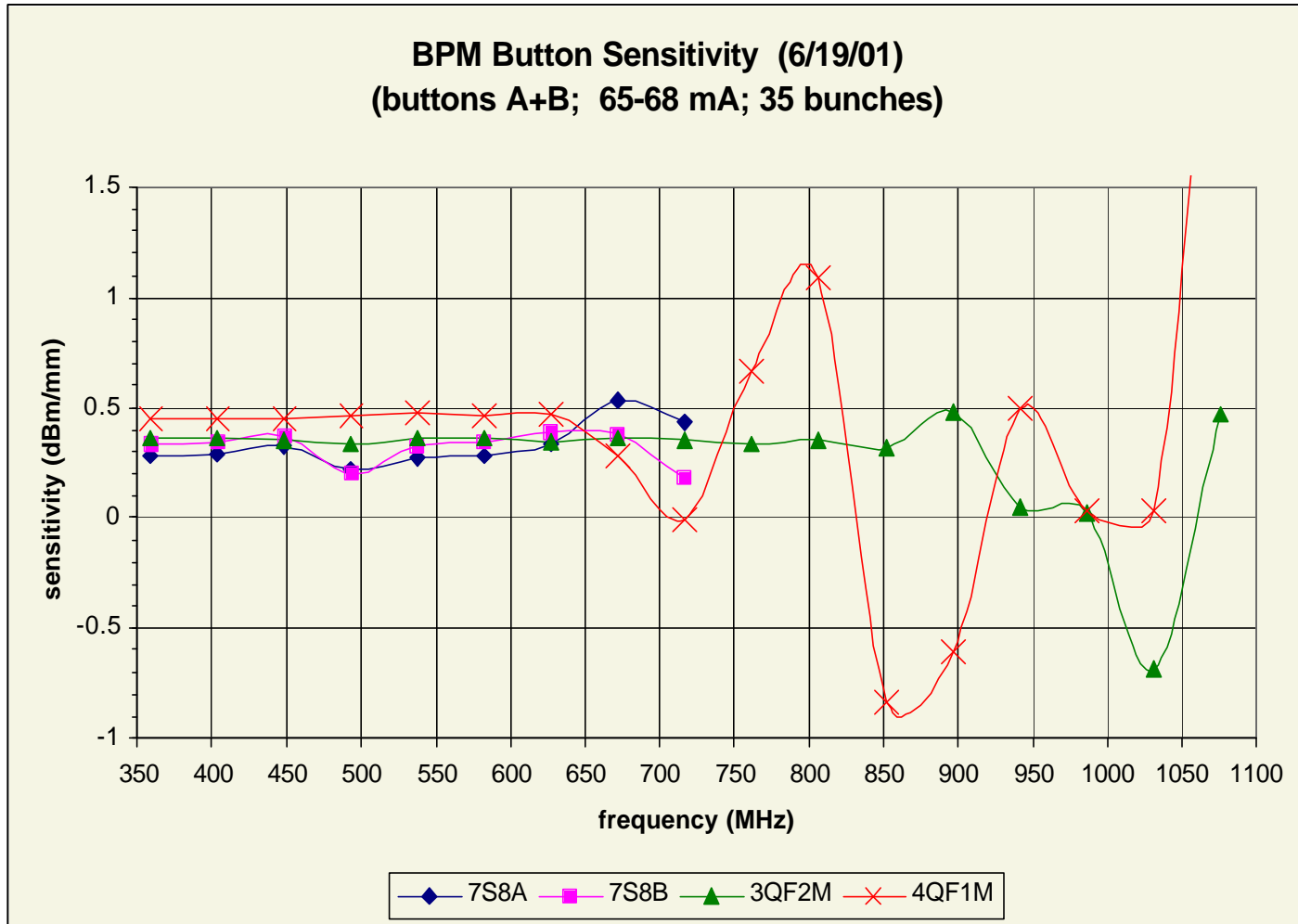
BPM locations

(104 total; 90 used for orbit feedback)

RF modes in SPEAR ID vacuum chambers



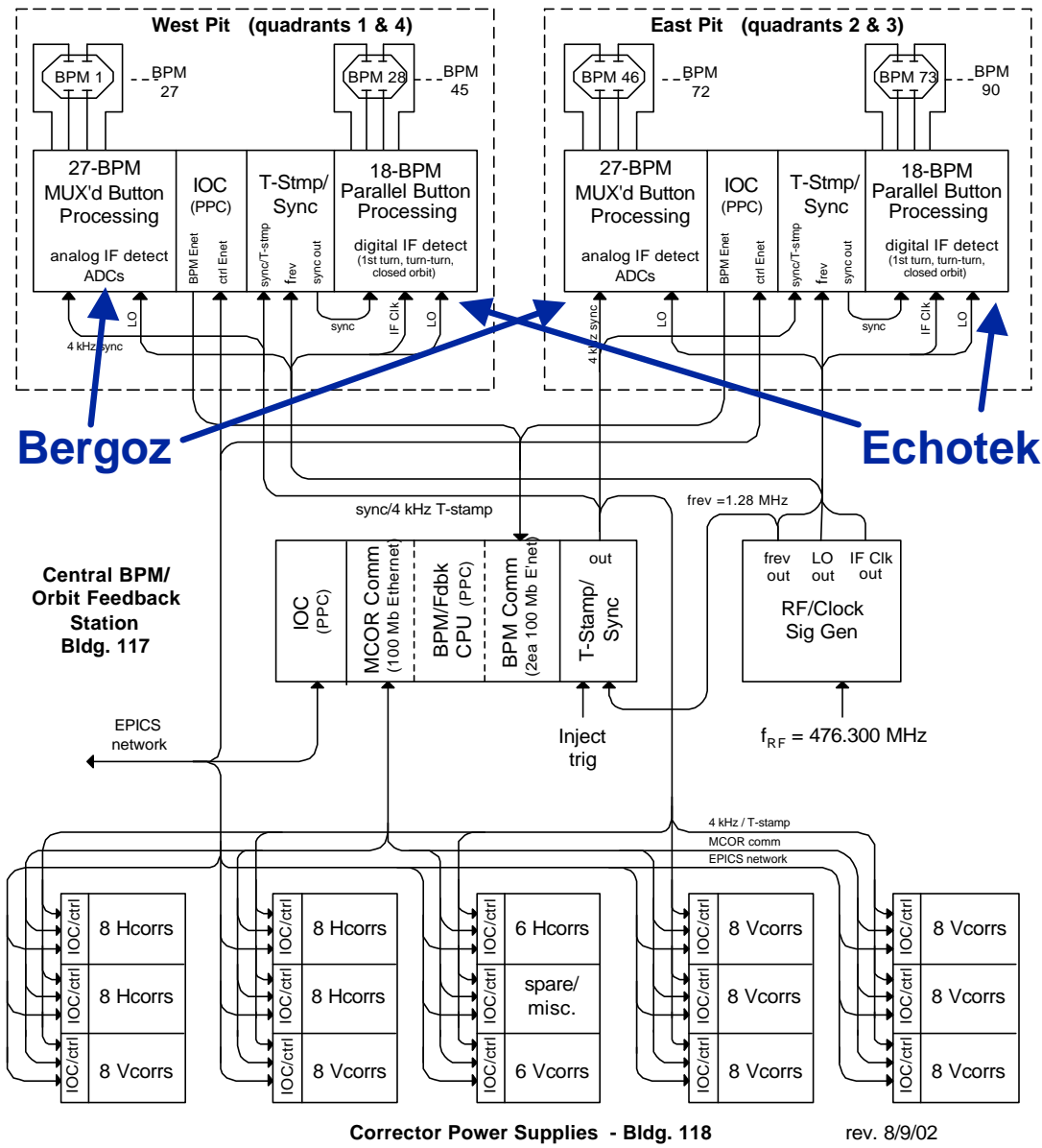
BPM gain vs. frequency



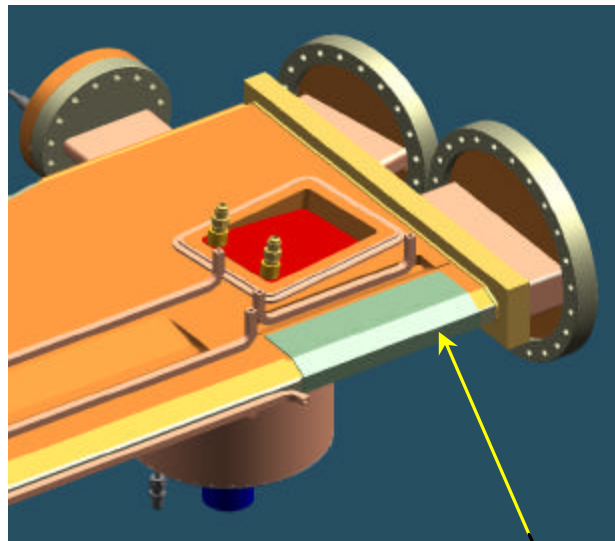


Digital orbit feedback

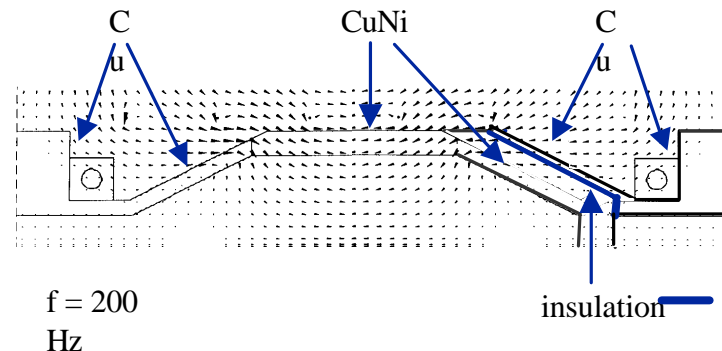
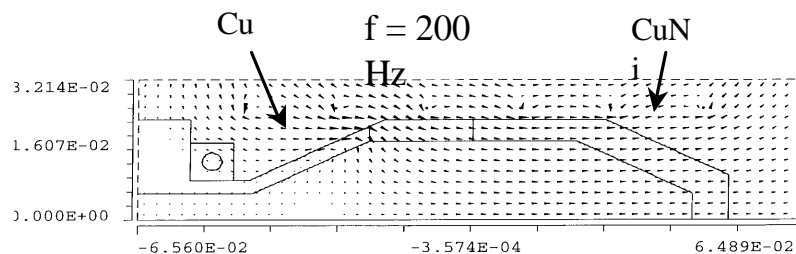
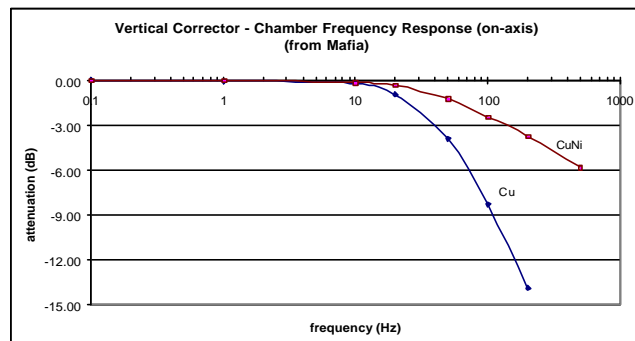
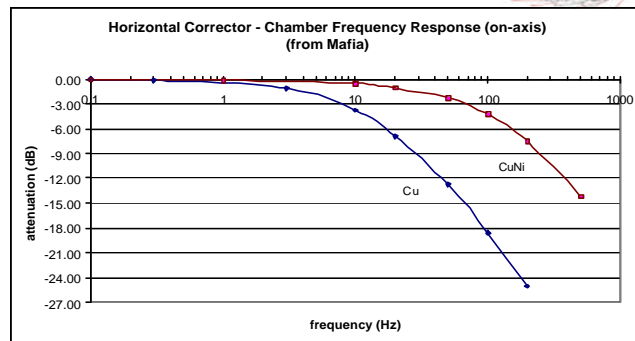
- 100 Hz bandwidth
- 4 kHz cycle time
- Electron + photon monitors
- ~18 effective bit steering magnets



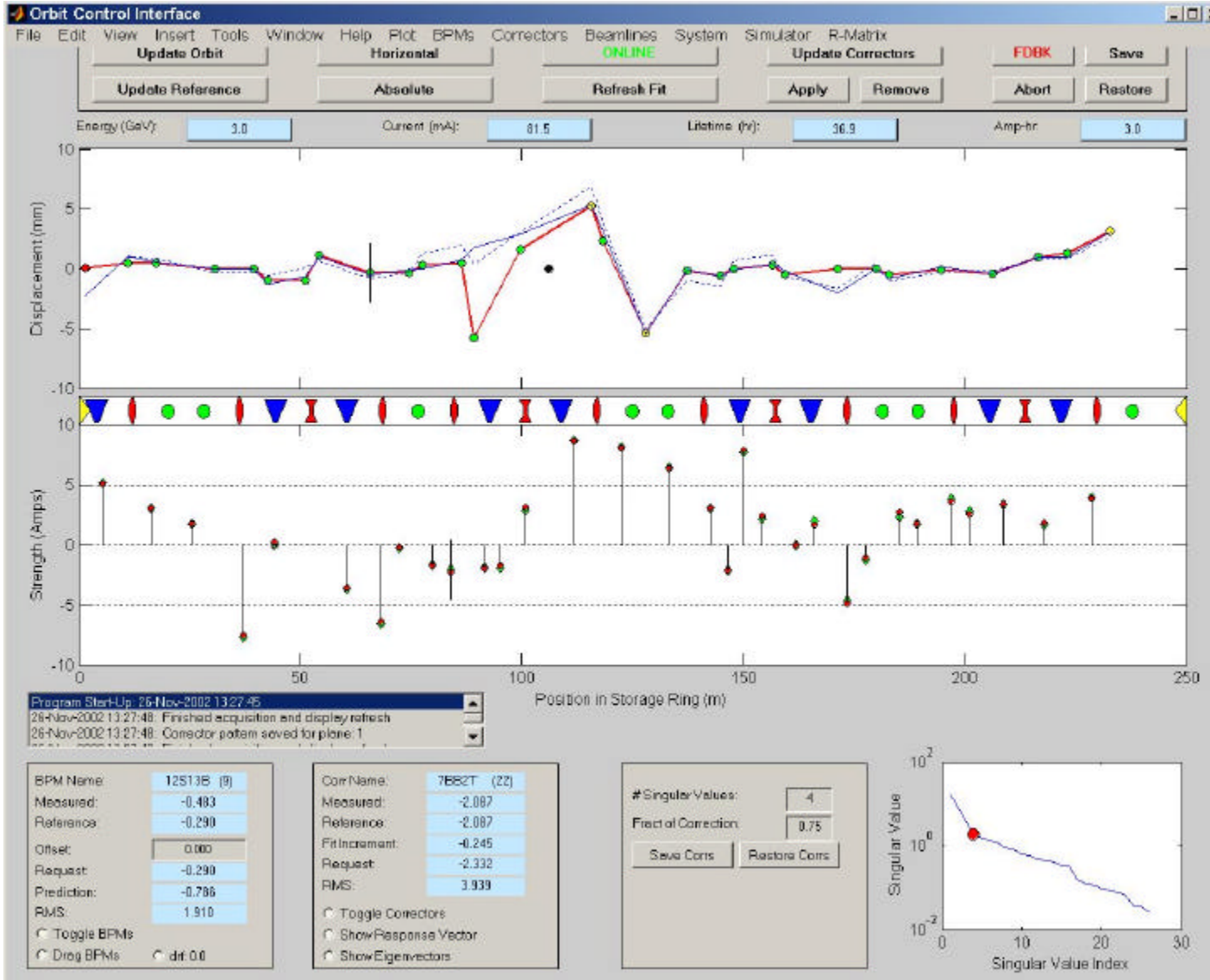
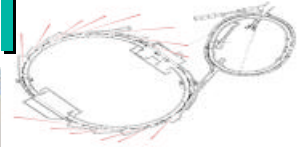
SPEAR3 Cu chamber eddy current break



CuNi

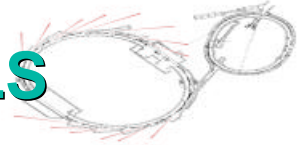


MATLAB-based orbit control



J. Corbett
A. Terebilo

Beam-based alignment algorithm from ALS



Quadrupole modulation:

GUI

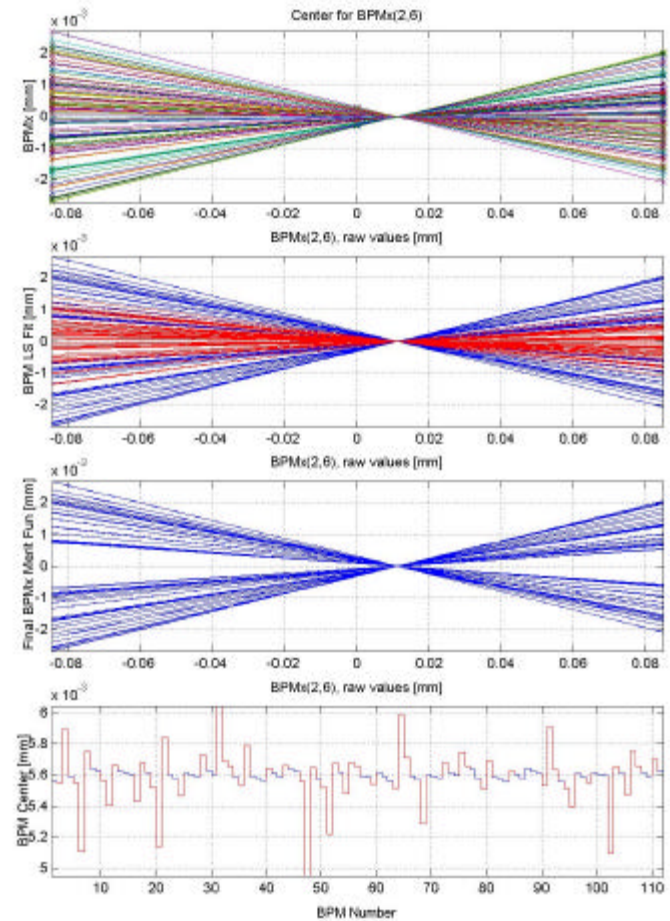
Simulation results

Sector	Quadrupole	Location	Phase	Centering Algorithm	
Sector 1	Q1	1	horizontal	Start	Abort
Sector 2	Q2	2	vertical		
Sector 3	QPC				
Sector 4					
Sector 5					
Sector 6					
Sector 7					
Sector 8					
Sector 9					
Sector 10					
Sector 11					
Sector 12					
Sector 13					
Sector 14					
Sector 15					
Sector 16					
Sector 17					
Sector 18					

Single Sheet	
OK	OK
Plot Old Data	
New Data Figure	

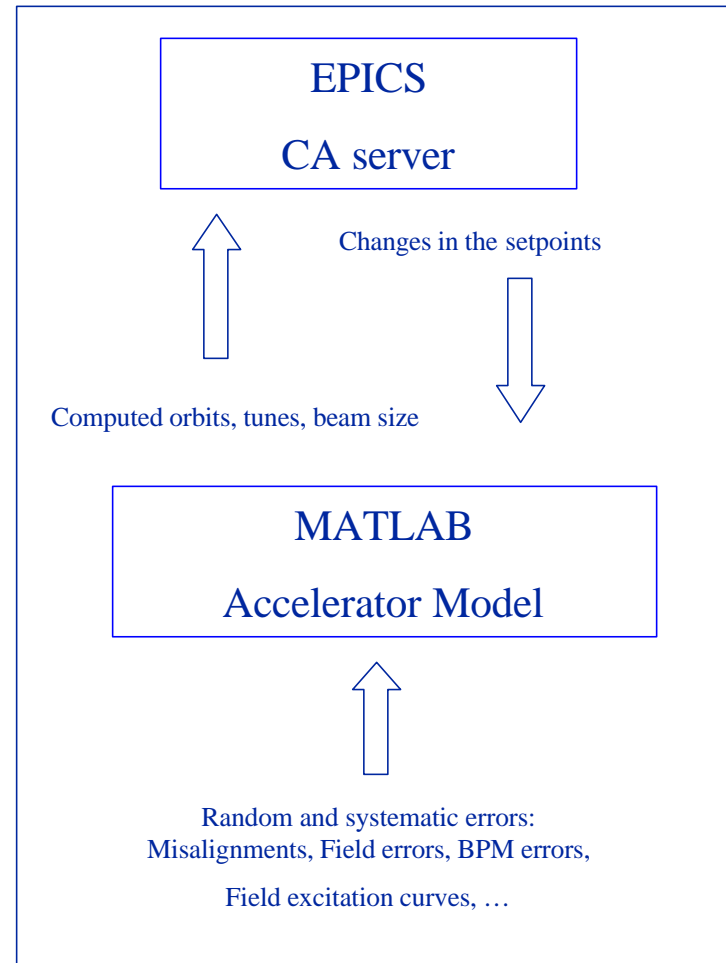
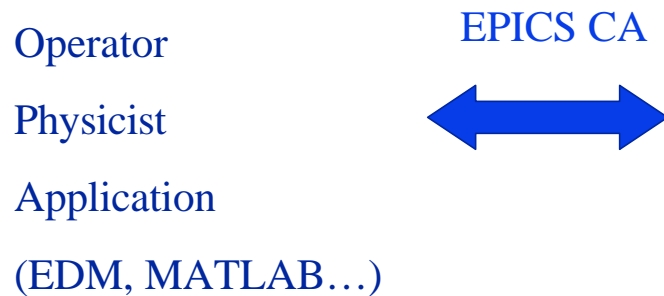
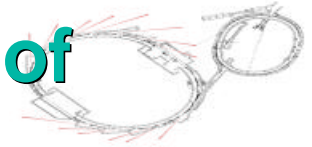
	Default Value	User Specified
Sheet Current	1	
Detector Noise	2000	
Connector Current	1	

Power Supply Currents		
	Quadrupole	Sheet
Initial	363.1523	363.1523
Fixed		
Delta		



Greg Portmann

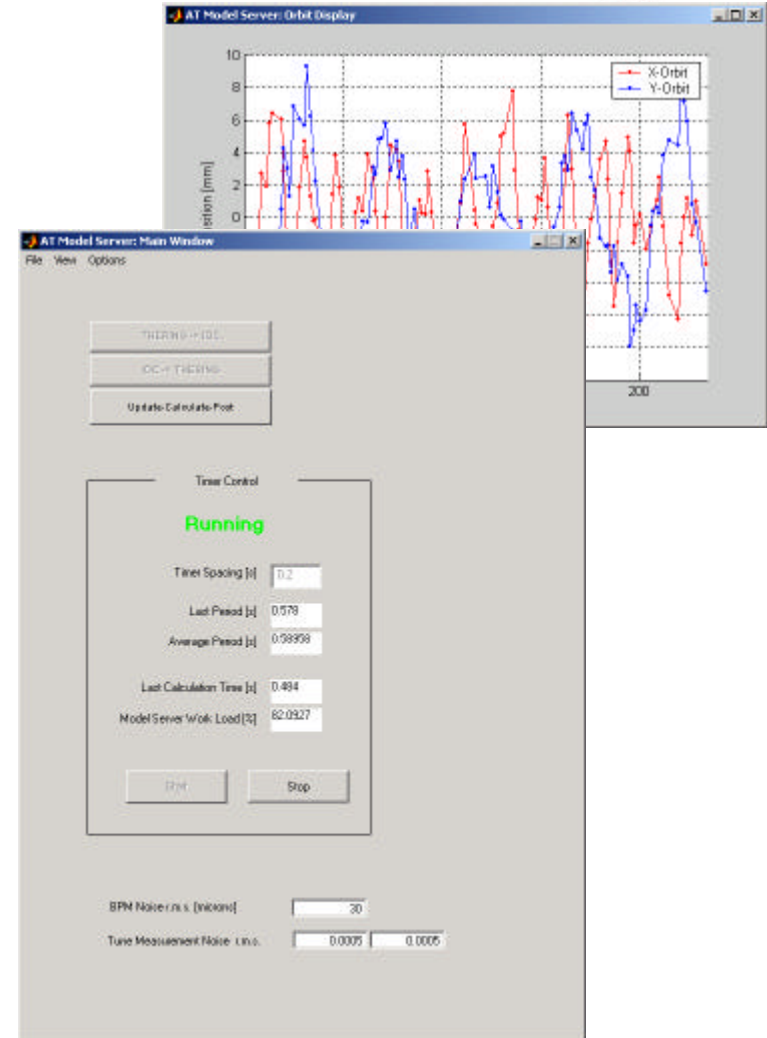
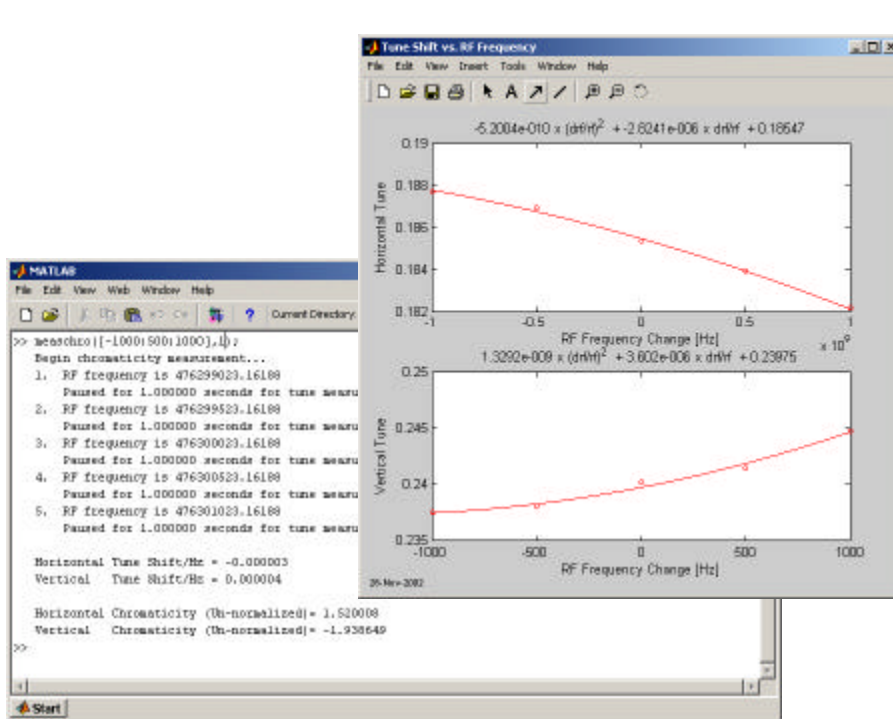
SPEAR3 simulated commissioning of application algorithms



A. Terebilo, J. Corbett

Beam Orbit Stabilization Workshop, Spring-8, December 4-6, 2002

SPEAR3 simulated commissioning



G. Portmann

Chromaticity measurement application (left)
running on an operators PC

tested against a

simulated SPEAR3 (right) running on a
dedicated PC

A. Terebilo

Acknowledgements



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